# Experimental and numerical study on the mechanical behaviour of CLT shearwalls with openings

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## 11 Abstract

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An investigation of the mechanical behaviour of CLT shearwalls, where either door or 12 window openings are cut out of the panel, is undertaken. The main aim of the study is 13 to investigate failure modes related to either mechanical anchor or CLT panel, based 14 on the geometrical dimensions and mechanical properties of shearwall. The results of 15 six full-scale monolithic CLT shearwalls with window or door openings are presented 16 and discussed. The results obtained from the full-scale shearwall tests are used to 17 validate a proposed numerical model, where input parameters, such as the mechanical 18 properties of the CLT panels and mechanical anchors, are obtained from component 19 20 level tests on beams and connections in isolation. The study shows that differently from single-panel shearwalls with no openings, brittle failure in the CLT panels is a possible 21 mode of failure, which needs to be considered in design. The failure mode in the CLT 22 panels is observed to occur either in bending or net shear in the lintel beams. The 23 proposed numerical procedure is found capable of estimating the maximum load with 24 reasonable accuracy, and the model predictions of the failure mode, number of centre 25 of rotations, and the overall deformation of the CLT panel are accurate for all the 26 studied specimens. 27

## 28 Keywords

Shearwalls; openings; timber structures; numerical models; beam tests; Cross
Laminated Timber.

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## 31 Highlights

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33	•	An investigation of the behaviour of CLT shearwalls with openings is
34		undertaken.
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36	٠	Failure modes related to either mechanical anchor or CLT panel are studied.
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38	٠	Full-scale shearwall tests were used to validate a proposed numerical model.
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40	•	Input parameters for numerical model were obtained from component level
41		tests.
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## 54 1. Introduction

The high strength-to-weight ratio and in-plane stiffness of Cross Laminated Timber 55 (CLT) panels, together with the ability of these structural systems to dissipate energy 56 through mechanical connections, have made them a valuable alternative to other 57 traditional materials, especially in seismic prone areas [1-2]. The appeal of using CLT 58 shearwalls lies in the relative simplicity of the procedure used in the design method, 59 where panels are assumed to possess superior (or infinite) in-plane stiffness, thereby 60 engaging the boundary connections through rigid-body rotation and/or translation. 61 Those boundary connections typically consist of relatively flexible and ductile fasteners 62 connecting adjacent panels together, and mechanical anchors (angle brackets and 63 hold-downs) that ensure the transfer of shear and overturning forces to lower storeys 64 or foundation. 65

The connections between panel elements may be omitted in design cases where 66 energy dissipation is not required (e.g. areas where wind loading governs design), and 67 alternatively, the entire shearwall could consist of a single CLT panel. This provides an 68 assembly that has very high stiffness and that is relatively easy to manufacture and 69 70 assemble on site. The need to have window and door openings in the walls necessitate cutting such openings directly in the panels, and thereby facilitating the assembly 71 process. Alternatively, the door and window spaces are accounted for during the 72 erection process and header beams and parapets are installed separately following 73 the installation of the wall segments. The difference in the behaviour, and consequently 74 the analysis procedure and design assumptions, between these two systems is 75 considerable. When the lintel beams and parapets are installed separately, it can 76 generally be assumed that the wall segments behave as cantilevers, the CLT panels 77 remain elastic, and the failure occurs in the mechanical anchors. Conversely, when the 78 openings are cut out of the CLT wall panel, the structural continuity between lintels and 79

wall segments is ensured and brittle failure in the CLT panels is a possible mode of 80 failure that designers need to consider. The prediction of this type of failure is 81 complicated by the presence of several factors, including variability in the wood 82 material, the multiplicity of possible failure modes and the high stress concentration 83 typically found at the edge of structural elements bordering the openings. The 84 variability found in wood material could lead to a diminished ability to predict the failure 85 mode with reasonable accuracy. Also, the difference between design level strength 86 and in-situ strength of wood element is not well established in order to predict the 87 sequence of failure between wood element and the mechanical anchors as well as 88 different failure modes possibly to occur within the wood element. Furthermore, the 89 theoretical stress level found near corners and edges of element may not always be 90 actually present in the physical element or may not initiate the failure due to the 91 reinforcing effects of the transverse laminates in CLT panels. 92

As will be demonstrated in the following section, several of the aforementioned issues 93 94 related to single wall panels with openings have not been addressed in the literature. The main motivation of the current study is to investigate the mechanical behaviour of 95 CLT shearwalls where either door or window openings are cut out of the panel. In 96 particular, the study aims to investigate failure modes related to either mechanical 97 anchor or CLT panel, based on the geometrical dimensions and mechanical properties 98 of shearwall. The link between the mechanical behaviour of CLT beams with vertical 99 outer layers and the behaviour of the CLT shearwall is also established through 100 experimental testing and numerical analysis using finite element (FE) model. 101

The methodology used in the current study involves experimental investigation of six full-scale monolithic CLT shearwalls with window or door openings. The geometrical dimensions of shearwalls, layout of the CLT panels, size of openings and type of mechanical anchors were selected with the intention of achieving a targeted failure mode related to either the lintel beam or the mechanical anchors. The results obtained from the full-scale shearwall tests were used to validate a numerical model, where input parameters such as the mechanical properties of the CLT panels and mechanical anchors were obtained from component level tests on beams and connections in isolation. The effects of vertical load and potential uplift stiffen

ss and strength of angle brackets have been omitted from this investigation in order to reduce the variables on the study's focus, which relates to the failure mechanism, especially in the lintel beam. The comprehensiveness of the proposed model and accompanying experimental campaign will be achieved by incrementally introducing such parameters in ongoing and future research effort.

116 2. State of the art

Establishing the behaviour of CLT shearwalls without openings has been the subject 117 of several research programmes in the past two decades. These studies have involved 118 119 significant experimental components of shearwall assemblies in isolation as well as part of a system at the building level. Analytical and numerical approaches have also 120 121 been developed to investigate the influence of mechanical connections on the structural performances of CLT buildings consisting of single- or multi-panel 122 shearwalls. The main outcome from the experimental testing was the confirmation that 123 the lateral behaviour of CLT shearwall without openings, at the ultimate limit state, is 124 governed by the mechanical performance of connections, while the CLT panels can 125 be assumed to behave almost elastically. 126

At the building level, Ceccotti et al. [3] performed shake-table tests on a 7-storey CLT building constructed with primarily multi-panel shearwalls. Tsuchimoto et al. [4] experimentally investigated the static and dynamic response of a 3-storey CLT structure with semi-rigid connections between wall segments and lintel beams demonstrating adequate seismic performance. Flastcher and Schickhofer [5]

investigated the seismic performance of a 3-storey CLT building with single-panel 132 shearwalls using a shake-table. The main finding from this study was that multi-panel 133 CLT shearwalls experienced more deformation but also more ductility than those 134 consisting of a single panel. Popovski and Gavric [6] studied the lateral behaviour of a 135 2-storey CLT building under monotonic and cyclic loading. The building was 136 137 characterized by single-panel CLT shearwalls with openings cut out of the panels in one direction, while multi-panels shearwalls were adopted along the other direction. 138 The failure was characterized by nail vielding in the brackets at the base of the wall 139 due to combined action of rocking and sliding. Significant slip along the vertical joints 140 in multi-panel shearwalls was detected. Van de Lindt et al. [7] performed shake-table 141 tests on a 2-storey CLT timber building in order to investigate the influence of panel 142 aspect ratios and presence of perpendicular CLT walls. The results showed that 143 shearwalls with high values of panel aspect ratio were governed by rocking failure, 144 while shearwalls with low values of panel aspect ratio were characterized mainly by 145 sliding mechanism. It was also observed that the perpendicular walls did not 146 significantly affect the rocking behaviour of shearwalls. Gavric and Popovski [32] 147 evaluated the influence of the perpendicular walls on the strength capacity of CLT 148 shearwalls through experimental cyclic tests on a 2-storey CLT house. The results 149 showed that the perpendicular walls increased the rocking strength capacity of 150 shearwall and altered the failure condition in most of the shearwalls to that of sliding. 151

A significant number of experimental studies have also been undertaken at the wall level primarily for CLT shearwalls without openings. Popovski et al. [8] conducted quasi-static tests on single- and multi-panel shearwalls, characterised by different wall aspect ratios. The study also investigated two-storey wall assemblies. The results from this investigation showed that the energy dissipation and ductility capacity of singlepanel shearwalls is related to the mechanical behaviour of hold-down and angle-

brackets. Hristovski et al. [9] performed shake-table tests on both single- and multi-158 panel CLT shearwalls showing that the mechanical anchors are able to dissipate 159 adequate seismic energy when a rocking behaviour is exhibited by shearwalls. Okabe 160 et al. [10] studied the influence of vertical load on the rocking behaviour of single- and 161 multiple-panel walls, concluding that vertical load can significantly increase the 162 strength capacity of CLT shearwalls. Gavric et al. [11] investigated the cyclic behaviour 163 of both single- and multi-panel shearwalls, reporting that the in-plane deformations of 164 the CLT panels were almost negligible and that the failure mode and inelastic 165 deformations were limited to the mechanical anchors and vertical joints. Akbas et al. 166 [12] studied the behaviour of self-centering CLT shearwalls connected to the 167 foundation by means of vertical post-tensioned steel bars. The study also provided 168 simple analytical expressions for the prediction of the lateral response of such 169 structural systems. Cyclic and monotonic tests were also conducted by Chen and 170 Popvoski [13] on balloon-type CLT sherwalls in order to validate a proposed 171 mechanics-based analytical model to predict the lateral response of such walls. The 172 experimental tests showed that coupled-panel balloon-type CLT shearwalls with semi-173 rigid and ductile vertical joints possess much larger plastic deformations than those 174 consisting of single-panel shearwalls. D'Arenzo et al. [14] investigated the lateral 175 behaviour of CLT shearwalls connected to the floor below by means of innovative bi-176 directional angle brackets. The study showed a comparable mechanical behaviour of 177 the tested shearwalls with those using traditional hold-down and angle brackets. 178

Research conducted on shearwalls with openings has been relatively limited and, in most cases, aimed only at defining reduction coefficients that take into account the effect of opening dimensions on the stiffness and strength capacity of shearwall. Dujic et al [15] presented the results of experimental and parametric numerical analyses of CLT shearwalls with different size and configuration of door and window openings.

Studies by Ceccotti et al. [16] and Flatscher et al. [17] presented results from 184 shearwalls with a door opening, where failure was observed in the mechanical anchors 185 used to connect shearwalls to the foundation, while the CLT panels behaved 186 elastically. Yasamura et al. [18] investigated a 2-storey CLT buildings constructed 187 using single-panel shearwalls with openings and reported observations of failure in the 188 mechanical anchors used to prevent the uplift of panels as well as the formation of 189 cracks at corners of the openings. The study emphasized the importance of 190 considering the panel failure in the design of CLT panels with openings. Pai et al. [19] 191 numerically investigated the force transfer around openings in CLT shearwalls and 192 identified the needs for local reinforcements to avoid premature failure in the panel. 193 Mestar et al. [20] established the kinematic modes of shearwalls with door or window 194 openings, based on the hold-down configuration and the geometrical dimensions of 195 the CLT panels. The experimental tests showed that failure mostly occurred in the 196 197 hold-down while bending failure in the CLT panel was observed in wall with door opening and high length-to-height aspect ratio of the lintel beam. 198

Most of the numerical studies on CLT shearwalls with openings aimed at determining 199 reduction coefficients to be applied to an equivalent CLT shearwall with no openings 200 [21-23]. Generally, 2-D area elements were implemented in the finite element models 201 to represent the behaviour of the CLT panels. An equivalent frame model was 202 proposed by Mestar et al. [24], as an alternative to the FE model with 2D area 203 elements, to establish the behaviour of CLT shearwalls with window or door openings. 204 Review of the available literature makes it clear that although a significant effort has 205 been made to establish the behaviour of CLT shearwalls with various geometrical 206 configurations and connection detailing, there is a clear gap in knowledge in relation 207 to the behaviour of monolithic shearwalls with openings. Studies on these structural 208 systems have been scarce and limited in scope to observations of various failure 209

modes, with little emphasis on failure occurring in the CLT panel. Such failure modes are naturally brittle and should be avoided, and hence need to be better understood. To the authors' knowledge, no experimental study has been undertaken with the aim to specifically achieve a targeted failure in the CLT panel. The current study aims to establish a better understanding of the behaviour of key parameters affecting the shearwall performance, such as lintel beams and mechanical anchors, in isolation as well as part of the wall assembly.

- 217
- 218 3. Experimental test set-up

In this section, the tests conducted on full-scale CLT shearwalls with openings and
those undertaken at the component level on CLT beams and mechanical anchors are
described.

#### 222 <u>3.1 Tests on shearwalls</u>

223 Monotonic tests were carried out on six CLT shearwalls with either door or window openings. The openings were cut out from CLT panels in order to maintain structural 224 continuity between the wall segments and the lintel beam and parapet. The panels 225 226 comprised of Spruce boards of C24 grade and width, w, of 170 mm, manufactured according to [33]. The total thickness,  $t_{tot}$ , of 3- and 5-ply panels were 90 mm and 100 227 mm, with layout of laminations of 30v-30h-30v and 20v-20h-20v-20h-20v, respectively. 228 The designation "*v*" and "*h*" here indicate the orientation of the lamination being vertical 229 and horizontal, respectively. 230

The wall height,  $h_{wall}$ , was equal to 2380 mm for all shearwall test specimens. Commercially available hold-down anchors (WHT620) were used to connect the wall to a steel base beam, representing the foundation. Each hold-down was connected to the wall panel using fifty-five 4x60 mm ring shanked nails, while the attachment to the steel base beam was achieved using an M20 bolt. Two different hold-down

configurations were adopted: a double hold-down configuration (DH), where hold-down
anchors were placed at both ends of each wall segments, and single hold-down
configuration (SH), where hold-downs were placed at the ends of the shearwall. The
choice of investigating these hold-down configurations was based on the results
obtained by Mestar et al. [20], which presented different kinematic behaviour of the
wall based on the hold-down configuration.

The geometrical dimensions and the opening layouts of the wall specimens are 242 presented in Table 1 and shown in Figure 1. As seen in Table 1, a label is presented 243 for each specimen, indicating the type of opening being a door or window (D or W), 244 number of lamination layers in the CLT panel (3 or 5), and whether the lintel is 245 considered to be relatively short or long with respective lengths of 600 mm or 900 mm 246 (S) and 1500 mm (L). Using this terminology, a specimen with label W 5 S would 247 consist of a window opening where the panels comprise of 5-ply CLT and a short lintel 248 beam. In Table 1, the variable  $I_{wall}$  represents the total length of the shearwall,  $I_{op}$  is the 249 length of the opening, while  $h_{lintel}$  and  $h_{par}$  are the height of the lintel and parapet, 250 respectively. Figure 2 also shows a photograph of the test setup using wall specimens 251 02 and 06 as examples. It can be noted in Figure 2 that the hold-down connection 252 farthest away from the load application point is always assumed to be subjected to a 253 compression force and has therefore been omitted. 254

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			n. of	$t_{tot}$	I <sub>wall</sub>	$h_{\textit{lintel}}$	h <sub>par</sub>	l <sub>op</sub>	Opening	Hold-down
	Test	Label	layers						type	config.
			[-]	[mm]	[mm]	[mm]	[mm]	[mm]	[-]	[-]
-	Wall 01	D_3_S	3	90	3300	340	-	600	Door	SH
	Wall 02	D_5_S	5	100	3300	340	-	900	Door	DH
	Wall 03	W_5_S	5	100	3300	340	340	900	Window	SH
	Wall 04	D_3_L	3	90	3900	510	-	1500	Door	DH
	Wall 05	D_5_L	5	100	3900	340	-	1500	Door	DH
	Wall 06	W 5 L	5	100	3900	340	340	1500	Window	DH

Table 1: layout of shearwall tests



**Figure 1:** Geometrical dimensions and the opening layout





**Figure 2**: wall 02 D\_5\_S (a) and wall 06 W\_5\_L (b)

Figure 3a provides important details on the blocking mechanism that was adopted in the test set-up in order to prevent the sliding of shearwall. The blocking consisted of a 15 mm thick steel plate, designed to prevent the development of localized high compression stresses in the wood, while a cylindrical steel section acted as a roller to allow free rotation of the wall. The lateral load was applied by means of a horizontal hydraulic jack, connected to a rigid steel frame. A 25 mm thick steel plate was used to transfer the load from the hydraulic jack to the top corner of the wall, as shown in Figure 3b. 



Figure 3: blocking mechanism at the bottom of the wall (a) and steel plate at the top of the
 wall (b)

275 Additional restraints were provided at the top of the wall specimen, on both sides, to prevent out-of-plane movement of the wall and buckling of the lintel beam (see Figure 276 2). The testing procedure was carried out in accordance with the EN594 standard [25]. 277 It is noteworthy to mention that the vertical load has been omitted in the current testing 278 program, in order to study the behaviour of the wall with opening using fewer 279 parameters. It can be expected that the vertical load would provide a stabilizing effect 280 281 on the rocking behaviour of the wall, while increasing the shear and bending forces in lintel elements, which in turn could lead to an increased probability of failure occurring 282 in CLT panels. It is recommended that future studies be carried out to investigate the 283 role and impact of the vertical load on the behaviour of the wall system. Figure 4 shows 284 the instrumentation layout to capture the various deformation contributions of the CLT 285 panel and the mechanical anchors. Two Linear Variable Displacements Transducers 286 (LVDTs) were used to measure the horizontal displacements  $\delta_{TOP,1}$  and  $\delta_{TOP,2}$  at the 287 top of shearwall at each end (LVDT 1 and 2), while another LVDT (either 3 or 4) was 288 positioned horizontally at the bottom of the wall to measure its sliding displacement 289  $\delta_{BOT}$ . Two additional LVDTs (5 and 6) were used to measure the uplift displacement  $V_1$ 290 and  $V_2$  at the bottom corner of the first wall segment (i.e. that closest to the load 291 application point and most prone to uplift). LVDTs 7 and 8 were connected to two 292 diagonal wires in order to measure the panel deformation  $d_T$  and  $d_C$ . The relative top 293

horizontal displacement of the wall,  $\delta$ , was obtained by calculating the difference between the top horizontal displacement measured by LVTD 1 or 2 (i.e.  $\delta_{TOP,1}$  or  $\delta_{TOP,2}$ ) and the sliding measured at the bottom of the wall,  $\delta_{BOT}$ .



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298 **Figure 4:** test measurement (front, a, and back, b)

#### 299 <u>3.2 Component-level tests</u>

As mentioned before, the purpose of conducting component level tests was to obtain a more accurate input parameters for the numerical model and to minimize the variability usually associated with failure in the wood material.

The CLT beam were selected from same batch as the CLT wall panels and they had 303 304 the same layup pattern, number of layers (3 and 5), orientation of laminates (Figure 5), species (spruce) as well as grade (C24), manufactured according to [33]. The 305 thickness of the beams and width of individual boards were selected such that they 306 were consistent with the full-scale shearwall specimens. The beam height, h, was 307 chosen to be a multiple of the board width, w, (i.e. h=2x170=340 mm). The specimen 308 configurations consisted of two different lengths, namely equal to 4.76 m and 2.72 m, 309 in order to promote bending and shear failure modes, respectively. 310



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**Figure 5:** cross section of beam specimen with 5- (a) and 3-layers (b)

The geometrical dimensions of the CLT beams and the number of specimens for each configuration are reported in Table 2, where the thickness and orientation of the individual laminations are also provided. Variables  $t_h$  and  $t_v$  represent the total thickness of laminations along the horizontal and vertical direction, respectively.

318	Table 2	<b>2:</b> layup	of specimens
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I <sub>beam</sub> = 4760 mm											
Specimen number	n. of specimen	n. of layers	Layup [mm]	t <sub>h</sub> [mm]	t <sub>v</sub> [mm]	t <sub>TOT</sub> [mm]					
B 01	3	5	20v-20h-20v-20h-20v	40	60	100					
B 02	3	3	30v-30h-30v	30	60	90					
		I <sub>beam</sub> :	= 2720 mm								
Specimen number	n. of specimen	n. of layers	Layup [mm]	t <sub>h</sub> [mm]	t <sub>v</sub> [mm]	t <sub>tot</sub> [mm]					
B 03	3	5	20v-20h-20v-20h-20v	40	60	100					
<b>B</b> 04	3	3	30v-30h-30v	30	60	90					

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Figure 6 shows the test setup and boundary conditions for the beam tests. Each beam was supported on steel rollers and loaded at the third points between the two bearing supports. The load was applied up to the specimen failure using a 250 kN hydraulic jack at a constant rate in accordance with EN408 [26]. Two LVDTs, one on each side of the beam, were used to measure the total vertical deflection,  $w_{glob}$ , at the mid-span, while two other LVDTs measured the local relative displacement,  $w_{loc}$ , between the

- 326 centre of the beam and a fixed point located in the null-shear zone in conformance with
- 327 EN408 [26].

329



Two monotonic tests were carried out on the same hold-down anchor that was adopted in the full-scale sherwall tests. The CLT specimens were loaded parallel to the direction of their outer layers. A symmetric layout of the test was ensured by connecting two hold-downs to each side of the specimen, as shown in Figure 7. The hold-downs were connected to the 400x700x100 mm CLT specimen using fifty-five 4x60 ring shanked nails. Four LVDTs (two one each side) were used to measure the vertical displacement of the CLT specimens relative to the base of the hold-down anchor.





- 338 **Figure 7:** tests on hold-down
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- 340 4. Experimental results and discussion
- 341 <u>4.1 Shearwall tests</u>

All shearwall specimens were loaded until failure was documented in either the holddown or lintel beam. The loading protocol persisted beyond potential initial cracks in the lintel beams until the ultimate failure was reached in order to detect any possible change in failure mode during the testing process. Failure in the hold-downs was characterised by a relatively brittle tensile failure in the steel plate along the bottom row of the nails, as shown in Figure 8.



349 **Figure 8:** hold-down failure in Wall 03 (W\_5\_S)

The failure in the lintel beam was characterised by a net shear or bending failure. The net shear failure was observed to occur at the end of the lintel section (i.e. near the wall segment) in all horizontal layers, as shown in Figure 9 for Wall 01 (D\_3\_S). The bending failure was observed to correspond to finger joints in the inner horizontal boards at the end of the lintel section, as shown in Figure 10 for Wall 04 (D\_3\_L).

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**Figure 9:** *failure in the lintel of Wall 01 (D\_3\_S)* 



**Figure 10:** *failure in the lintel of Wall 04 (D\_3\_L)* 

The load-displacement curves for all the tested walls are shown in Figure 11, where 360 the point at which the hold-down anchor (HD) or the CLT lintel beam fails are indicated. 361 In general, it can be observed that the behaviour is brittle since it is influenced by failure 362 in the hold-down or the lintel beam, both of which have brittle behaviour. Although the 363 failure in the CLT panel is expected to be brittle, the behaviour of the hold-down 364 depends on their nailing configuration. In the current study, and in order to observe 365 failure in the lintel beam, commercially available fully-nailed hold downs were adopted, 366 which led to the brittle failure observed. Other hold-downs may possess more ductility 367 (e.g. partially nailed) and therefore prioritizing failure in the hold-down rather than the 368 CLT panel is preferred in design because the overall behaviour of the wall assembly 369 could be better controlled. Failure in the CLT panel will always be brittle and therefore 370 it should be avoided when possible. 371





373

**Figure 11:** force vs displacement curves of full scale shearwall specimens.

Experimental results, including maximum load,  $F_{max}$ , and corresponding displacement,  $\delta_{F_{max}}$ , load and displacement corresponding to the hold-down failure,  $F_{u,hd}$  and  $\delta_{u,hd}$ ,

- load and the displacement corresponding to failure in the CLT panel,  $F_{u,CLT}$  and  $\delta_{u,CLT}$ ,
- and the lateral stiffness of the shearwall, k, calculated as the slope between 10% and
- 40% of the maximum load, are reported in Table 3.

Test	Label	Failura mode	F <sub>max</sub>	$\delta_{F_{max}}$	F <sub>u,CLT</sub>	$\delta_{u,CLT}$	F <sub>u,hd</sub>	$\delta_{u,hd}$	k
Test	Laber	I allule mode	[kN]	[mm]	[kN]	[mm]	[kN]	[mm]	[kN/mm]
Wall 01	D_3_S	CLT panel	113	18.4	106	19.7	-	-	9.54
Wall 02	D_5_S	Hold-down	176	24.3	-	-	174	24.8	13.71
Wall 03	W_5_S	Hold-down	161	18.1	-	-	160	18.3	12.58
Wall 04	D_3_L	CLT panel	179	26.8	179	26.8	-	-	10.25
Wall 05	D_5_L	Hold-down	166	27.7	-	-	166	27.7	8.64
Wall 06	W_5_L	Hold-down	227	29.4	-	-	227	29.4	9.66

**Table 3:** mechanical parameter obtained from the force-displacement curve

With the exception of Wall 03 (specimen W 5 S), the deformed shapes for all the other 383 tested shearwalls were characterized by two centres of rotation, one at each wall 384 segment, as shown in Figure 12a for Wall 01 (D\_3\_S). This observation was possible 385 due to the monitoring of the uplift displacement,  $v_1$ , which had a relatively small 386 negative value, implying compression forces at that location. The bottom edge of the 387 shearwall at the opposite end was, as expected, always in compression. Wall 03 388 exhibited a kinematic mode consistent with the other shearwalls initially, however after 389 390 10 mm of horizontal displacement, the wall behaviour shifted to that of single centre of rotation, , as shown in Figure 12b, which coincided with the yielding of the nails in the 391 hold-down. 392



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Figure 12: uplifts in outermost wall segments for Wall 01 (a) and Wall 03 (b)

## 397 <u>4.2 Component-level tests</u>

As anticipated, all beam specimens with length equal to  $l_{beam} = 4760$  mm (i.e. B 01 and B 02), failed in bending, and the failure occurred near the mid-span of the beam elements, as shown in Figure 13. The failure mode in the beam specimens with length equal to  $l_{beam} = 2720$  mm (i.e. B03 and B04) was less consistent and involved bending failure in one of the specimens (#1 for B03 and #1 for B04), while net shear failures in the two horizontal laminations was observed in the other two specimens (#2 and #3 for B03, #2 and #3 for B04), as shown in Figures 14 and 15.











**Figure 15:** tests on short beams B 04 (specimen #1), a, and B 04 (specimen #3), b.

- Table 4 presents the results obtained from the beam tests, including maximum load,
- $F_{max}$ , and the corresponding failure.

## **Table 4:** *maximum load and failure mode of CLT beam tests*

Configuration d	and Tes	$F_{max}$ [kN]	Failure mode	
		#1	61.1	Bending
	B 01	#2	52.3	Bending
lhoom = 4760 mm		#3	57.3	Bending
		#1	60.0	Bending
	B 02	#2	19.5	Bending
		#3	44.3	Bending
		#1	172.9	Bending
	B 03	#2	180.8	Net shear
lhoom = 2720 mm		#3	217.4	Net shear
		#1	111.7	Bending
	B 04	#2	118.2	Net shear
		#3	128.8	Net shear

Table 5 provides the effective local modulus of elasticity along the major direction,  $E_l$ , calculated based on a linear regression between the 10% and 40% of the maximum load, according to EN408 [26], and considering only the contribution of horizontal layers when calculating the area moment of inertia  $I_{net}$ , as presented in Equation (1):

420 
$$I_{net} = \frac{t_h \cdot h^3}{12}$$
 (1)

The bending strength of the beams,  $f_m$ , obtained from the maximum load on beams with lengths  $l_{beam} = 4760$  mm, was calculated using Equation 2.

$$423 f_m = \frac{M_{max}}{W_{net,h}} (2)$$

424 where the maximum bending moment,  $M_{max}$ , and the elastic section modulus,  $W_{net}$ , 425 are calculated, as shown in Equation (3) and (4), respectively:

$$426 \qquad M_{max} = \frac{F_{max} \cdot a}{2} \tag{3}$$

427 
$$W_{net} = \frac{t_h \cdot h^2}{6}$$
 (4)

where *a* is the distance between the support and the load application point, equalling1360 mm.

It can be noted that the variability for beam B 02 is very high, even though the mean value obtained is consistent with the expected average bending strength. Since only six tests have been conducted and the current study is one of the first of its kind to address the behaviour of CLT beams with vertical outer laminations, it cannot be determined with certainty whether some of the values obtained represent outliers. All data points are presented here to allow future studies by the authors and others to evaluate this observation further.

Test			$E_l$	Mean	CoV	$f_m$	Mean	CoV
			[MPa]	[MPa]	[-]	[MPa]	[MPa]	[-]
		#1	14071			53.9		
	B 01	#2	14006	13878	2%	46.1	50.2	8%
$l_{hoom} = 4760 \text{ mm}$		#3	13558			50.5		
		#1	14286			70.6		
	B 02	#2	12062	13411	9%	22.9	48.5	50%
		#3	13886			52.1		

From the tests on beams with lengths  $I_{beam} = 2720$  mm, the net shear strength capacity  $f_v$  was calculated and presented in Table 6 by assuming a parabolic distribution of the internal shear stress according to the Jourawski theory [31], as presented in Equation (5).

$$444 f_v = \frac{3}{2} \cdot \frac{\frac{F_{max}}{2}}{t_{net} \cdot h} (5)$$

445 where  $t_{net} = \min(t_h; t_v)$ .

The torsional shear strength capacity  $f_T$  was not determined since no failure due to torsional shear between lamination was observed in the tests. When a shear failure mode did not occur in the shear tests,  $f_v$  represents the lower bound value of the net shear strength capacity.

450	Table	6:	net	shear	strenati	h

Test			Failure mode	$f_v$
		#1	Bending	>9.3
	B 03	#2	Net shear	10.0
lbeem = 2720 mm		#3	Net shear	12.0
		#1	Bending	>8.2
	B 04	#2	Net shear	8.7
		#3	Net shear	9.5

A relatively brittle failure was observed in the net section of the hold down steel plate. The maximum load,  $F_{max}$ , and the corresponding displacement,  $v_{Fmax}$ , as well as the lateral stiffness,  $k_{el}$ , are reported in Table 7 and the average load-displacement curve obtained from the tests is shown in Figure 16.

456 **Table 7:** hold-down elastic stiffness and ultimate load and displacement



457

458

459 **Figure 16:** *load-displacement average curve from tensile tests on hold-down* 

460 **5.** Numerical analysis

## 461 <u>5.1 Description of the model</u>

Numerical models were developed in the software package SAP2000 [27] to simulate the mechanical behaviour and failure modes of the tested CLT shearwalls. The methodology used in the model is consistent with that reported in [14,20]. Four-joints quadrilateral homogeneous shell elements with a mesh size equal to 37.5 x 37.5 mm were adopted for the modelling of the in-plane behaviour of the CLT panels. An example of the numerical model for Wall 02 (D\_5\_S) can be seen in Figure 17. The thickness of the shell elements was taken equal to the thickness of the wall CLT panel. Linear elastic orthotropic material properties were assigned to the shell elements. Effective modulus of elasticity  $E_{eff,h}$  and  $E_{eff,v}$  for the two in-plane directions were defined to take into account the different CLT panel layups as expressed by Equations 6 and 7:

473 
$$E_{eff,h} = \frac{E_0 \cdot t_h}{t_{tot}}$$
 along the horizontal direction (6)  
474  $E_{eff,v} = \frac{E_0 \cdot t_v}{t_{tot}}$  along the vertical direction (7)

475

where  $E_0$  is the mean value of modulus of elasticity parallel to the grain obtained from the beam tests.  $t_{tot}$  is the total thickness of the panel, while  $t_h$  and  $t_v$  represent is the total thickness of the horizontal and vertical layers, respectively. An effective in-plane shear modulus,  $G_{eff}$ , was determined according to the Equation 8, which takes into account both shear and torsional deformations of the laminations, as proposed by Brandner et al. [28].

$$482 \qquad G_{eff} = \frac{G_0}{1 + 6 \cdot \alpha_T \cdot \left(\frac{t_{mean}}{w}\right)^2} \tag{8}$$

483

where  $G_0$  is the shear modulus of the laminations, obtained from EN 338 [29], *w* is the width of laminations,  $t_{mean}$  is the mean thickness of laminations, calculated according to Equation 9, and  $\alpha_T$  is obtain using Equation 10.

$$487 t_{mean} = \frac{t_{tot}}{N} (9)$$

$$488 \alpha_T = p \cdot \left(\frac{t_{mean}}{w}\right)^q (10)$$

490 where *N* is the number of layers, *q* is equal to -0.79 and *p* is equal to 0.53 and 0.43 for

491 3 and 5 layers of the CLT panel, respectively, as reported in [28]. The values of  $E_{eff,h}$ ,

492  $E_{eff,v}$  and  $G_{eff}$  are reported in Table 8 for each panel layups.

Ν E<sub>eff,h</sub> t<sub>tot</sub> layup  $E_0$  $G_0$  $t_h$  $t_v$  $E_{eff,v}$ G<sub>eff</sub> [MPa] [-] [MPa] [MPa] [MPa] [mm] [-] [mm] [mm] [MPa] 100 20v-20h-20v-20h-20v 5551 8327 5 13878 690 40 60 578 3 30v-30h-30v 4470 8940 90 13411 690 30 60 494

493 **Table 8:** values of equivalent modulus for each layup of CLT panels



494



Figure 17: FE numerical model for Wall 02 (D\_5\_S)

497

Each hold-down was modelled using 1-joint multi-linear elastic link element with 498 mechanical behaviour represented by the average load-displacement curve obtained 499 from the two tensile tests reported in Section 4. Rigid compression-only (i.e. gap) 500 501 elements, located along the base of the shearwall, were used to simulate the contact between the CLT panels and the steel base beam. Rigid horizontal restraints were 502 applied at bottom corners of the wall segments in a manner consistent with those found 503 in the wall tests, as shown in Figure 3. A displacement-controlled non-linear static 504 analysis was performed by increasing the lateral displacement at the top of the 505 shearwall. 506

## 508 <u>5.2 Prediction Investigation of failure modes</u>

The failure condition related to the CLT panels was determined by means of a stepby-step verification of the axial ( $n_h$  and  $n_v$ ) and shear (v) internal forces per unit length in the shell elements, according to Equation 11.

512 
$$n_h = n_{R,h}; n_v = n_{R,v}; v = v_R$$
 (11)

where  $n_{R,h}$ ,  $n_{R,v}$  are the axial strength per unit length in the horizontal and vertical directions, respectively, and  $v_R$  is the shear strength per unit length. The axial strength per unit length can be calculated according to Equations 12 and 13 as the product of either the tensile,  $f_{t,CLT}$ , or compressive,  $f_{c,CLT}$ , strength of laminations and the total thickness along the horizontal,  $t_h$ , and vertical,  $t_v$ , direction, respectively.

518 
$$n_{R,h} = \begin{cases} f_{t,CLT} \cdot t_h, & \text{if } n_h \ge 0 \\ f_{c,CLT} \cdot t_h, & \text{if } n_h < 0 \end{cases}$$
 (12)

519 
$$n_{R,\nu} = \begin{cases} f_{t,CLT} \cdot t_{\nu}, & \text{if } n_{\nu} \ge 0 \\ f_{c,CLT} \cdot t_{\nu}, & \text{if } n_{\nu} < 0 \end{cases}$$
 (13)

520 The net shear strength,  $v_{R,net}$ , and the torsional shear strength,  $v_{R,tor}$ , can be obtained 521 as expressed by Equations 14 and 15:

$$522 \quad v_{R,net} = f_v \cdot min(t_h, t_v) \tag{14}$$

523 
$$v_{R,tor} = min\left(\frac{f_{tor} \cdot w \cdot t_v \cdot n_{CA,i}}{3 \cdot t_{v,i}}\right)$$
 for i=1, 3 and 5 (15)

where  $f_v$  is the mean value of the net shear strength obtained from the beam tests,  $n_{CA,i}$ is the number of crossing area that the i-th vertical layer shares with adjacent layers,  $f_{tor}$  is the mean value of the torsional strength and  $t_{v,i}$  is the thickness of the *i-th* vertical layer.

It is noteworthy to mention that the variables  $n_{R,h}$  and  $n_{R,v}$  represent the strength 528 capacities of the CLT panel subjected to a pure compressive or tensile force, which acts 529 uniformly along the entire section. In the case being investigated, the CLT elements are 530 subjected primarily to bending, and as such, the bending strength capacity is more 531 appropriate in the determination of strength capacity of the element [34]. Also, since the 532 outputs from shell elements are expressed in terms of axial internal forces per unit 533 length,  $f_{c,CLT}$  and  $f_{t,CLT}$ , in Equations 12 and 13 have been replaced with the bending 534 strength,  $f_{m.CLT}$ , obtained from the beam tests presented in Section 2. The axial strength 535 per unit length can hence be calculated as expressed by Equations 16 and 17. 536

537 
$$n_{R,h} = f_{m,CLT} \cdot t_h$$
 (16)

538  $n_{R,v} = f_{m,CLT} \cdot t_v$  (17)

It was observed from the experimental tests on CLT beams that the shear failure mechanism was related to a net shear failure. This is likely due to the values of widthto-thickness ratio adopted in the experimental campaign, which ensured that torsional shear failure mechanism was supressed. The axial and shear strength per unit length, calculated according to the equations presented in this section, are reported in Table 9 for each panel layup.

545 **Table 9:** axial and shear strength per unit length of the CLT panels

t <sub>tot</sub>	Ν	layup	$f_{m,CLT}$	$f_{v,CLT}$	$t_h$	$t_v$	$n_{R,h}$	$n_{R,v}$	$v_R$
[mm]	[-]	[-]	[MPa]	[MPa]	[mm]	[mm]	[kN/m]	[kN/m]	[kN/m]
100	5	20v-20h-20v-20h-20v	50.20	10.96	40	60	2008	3012	438
90	3	30v-30h-30v	48.52	9.08	30	60	1456	2911	272
16									

<sup>546</sup> 

547 As mentioned before, the magnitude of stress obtained from the numerical model near 548 the corner zones is not representative of the local real stresses. The values given by 549 numerical model are notoriously much higher than the values expected to be 550 experienced by the physical test specimen. This phenomenon has also been highlighted and discussed by other studies dealing with interpretation of numerical data [30]. For this reason, the verifications were performed by excluding a distance from the edge of the CLT panel, equal to the mesh size (37.5 mm), which corresponds approximately to 10% of the height of the section. Although, as will be presented in the next section, this approach seems to provide accurate and realistic predictions of the internal stresses, further studies are needed to investigate the internal stress distribution.

#### 557 5.3 Validation of the FE models

The validation of the proposed procedure in the numerical models was carried out by comparing the results obtained from the analyses with those from shearwall tests in terms of failure modes, load-displacement curves, number of centre of rotations as well as deformation in the CLT panels.

The comparisons between the load-displacement curves obtained from the test results 562 563 and FE models are presented in Figure 18. Additionally, numerical comparisons in terms of wall stiffness, K, maximum shear force,  $F_{max}$ , mode of failure and number of centres 564 of rotation (CoR) at ultimate displacement are reported in Table 10. The percentage 565 difference  $\varepsilon$  is calculated and reported for the stiffness and strength values. Also 566 provided in Table 10 is the reserved capacity (i.e. overcapacity - OC) of the component 567 (hold-down or CLT panel) that did not govern the failure of the wall, obtained from the 568 FE models. 569

		<i>K</i> [kN/mm]		F <sub>max</sub> [kN]		failure mode		CoR		OC of unfailed component			
Test	Label	FE	test	ɛ [%]	FE	test	ɛ [%]	FE	test	FE	test	Component	OC [-]
Wall 01	D_3_S	7.8	9.5	-17.7	97.8	112.8	-13.3	CLT shear	CLT shear	2	2	HD	1.20
Wall 02	D_5_S	9.1	13.7	-33.7	182.2	176.5	3.2	HD	HD	2	2	CLT	1.02
Wall 03	W_5_S	9.5	12.8	-25.9	136.0	160.9	-15.5	HD	HD	1	1	CLT	1.79
Wall 04	D_L_L	9.1	10.2	-11.1	171.1	178.6	-4.2	CLT shear	CLT bending	2	2	HD	1.03
Wall 05	D_5_L	8.6	8.6	0.1	162.0	165.9	-2.4	HD	HD	2	2	CLT	1.18
Wall 06	W_5_L	9.8	9.7	1.8	240.0	226.8	5.8	HD	HD	2	2	CLT	1.01

#### 570 **Table 10:** comparison between FE analyses and experimental tests

From Table 10, it can be observed that the maximum discrepancy between the model 571 and test results regarding the maximum load is 15.5%. This is considered reasonable 572 provided the variability found in wood and hold-down connectors. The comparison of the 573 initial stiffness shows more variability, which is expected since stiffness is notoriously 574 more difficult to estimate. The model prediction of the failure mode is accurate for all the 575 studied specimens, which is an important and encouraging finding as it presents another 576 evidence of the appropriateness of the proposed modelling procedure. Similarly, the 577 578 model was capable of correctly predicting the number of centre of rotations at the ultimate condition. It is noteworthy to mention that for shearwalls 01, 03 and 05 the 579 overcapacity of the unfailed component is quite large, indicating that a clear failure mode 580 was obtained from the models. Conversely, for shearwalls 02, 04 and 06, the values of 581 overcapacity of the unfailed component is close to unity, showing a more balanced 582 failure mode between CLT panel and hold-down. 583

The comparison presented in Figure 18 shows that the prediction of the model is quite reasonable. In general, predicting the overall behaviour and failure point in wall specimens, where the failure occurred in the CLT lintel beam (Walls 01 and 04), seems less accurate than when the failure occurred in the hold-down anchors (Walls 02, 03, 05 and 06). This is expected since less variability is associated with failure in the steel bracket in the hold-down.



591 Figure 18: comparison between the load-displacement curves of FE model and experimental test592

It should be noted that the distribution of the internal horizontal axial and shear forces were obtained at the points where the values of internal forces exceeded the strength values,  $n_{R,h}$  and  $v_R$  (Equations 14, 15 and 17), as shown in Figures 19 for Wall 01 and 04, respectively. The values are selected at the analysis step corresponding to the failure point represented in Figure 18.



598

**Figure 19:** *distribution of shear forces greater than the corresponding strength capacities detected at the ultimate condition for Wall 01 (a) and Wall 04 (b).* 

601

The displacements measured by the two diagonal LVTDs attached to the CLT panel in the experimental tests were compared to those obtained from the FE model, as shown

in Figure 20. The values from the FE model were obtained by determining the relative 604 displacement between the joints where the diagonal LVTDs were attached on the 605 shearwall. It can be seen that a good accuracy was obtained, showing the reliability of 606 the methodology used to model the CLT panels by means of homogenous shell 607 elements with effective moduli of elasticity and effective shear modulus. It is particularly 608 noteworthy to mention that the proposed methodology is adequate even when 609 significant deformations are observed in the CLT panels due to openings. In general, 610 since relatively small deformations are observed in shearwalls without openings, even 611 significant deviations in estimating the panel deformation have little effect on the overall 612 prediction. Contrarily, the flexibility of the lintel beam in shearwall with openings leads 613 to an overall panel flexibility that cannot be ignored in the analysis. As such, the obtained 614 results in the study related to the deformation along the diagonals of the shearwall, can 615 also be considered novel and further emphasizes the adequacy of the proposed model. 616



## 621 6. Conclusions

In this study, the mechanical behaviour of CLT shearwalls, where either door or window openings are cut out of the panels, is investigated through full-scale experimental tests and numerical analyses. The main conclusions that can be drawn from the current study are:

- experimental tests showed that failure mode of CLT shearwalls with openings
   can occur either in mechanical anchors or in the CLT panels, depending on the
   geometrical dimensions and mechanical properties of the shearwalls. Differently
   from single-panel shearwalls with no openings, the brittle failure in the CLT
   panels is a possible mode of failure that designers need to consider;
- the failure mode in the CLT panels was observed to occur either in bending or
   net shear in the lintel beam, depending on the layup pattern, number of layers,
   orientation of laminates and the geometrical and mechanical properties of the
   shearwall. Although based on the current investigation failure mode in wall
   segments and parapets seems less likely, potential future work should consider
   such failure modes;
- the proposed numerical procedure was capable of predicting the maximum load
  with reasonable accuracy, provided the variability found in wood and hold-down
  connectors. The model prediction of the failure mode, number of centre of
  rotations, and the overall deformation of the CLT panel was accurate for all the
  studied specimens.
- 642
- 643
- 644
- 645

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